

# **CONDUCTIVITY STUDY**

# Conductivity Study on Selected Aluminium Tubes/Ducts identified by Bob Swaim (to go to NTSB Lab on 10/9/96)

Using a Sigma scope SMP1

Conductivity measurements were taken on the tubes <sup>conductivity</sup> probe to determine if their temper was altered due to fire damage during the accident. The tubes are either fabricated from 6061-T4 or T6 Aluminium. However, since not all part numbers could be identified an exact material could not always be confirmed. Table I from BAC 5946 "Temper Inspection of Aluminium Alloys" gives an acceptable conductivity range for 6061-T4 as 36.0-45.5% IACS and 40.0-51.0 for 6061-T6. When material is thinner than .063 inch a correction factor must be applied to readings (see Table XIII from BSS 7351). Material that has been heated to a point which can cause brittle fracture under load (exceeding the solution treatment temperature) will result in conductivity values usually in the low to mid 30's %IACS. Material that has been heated above its normal aging temperature but below the solution treatment temperature will result in an increase in conductivity. The following results were obtained from the tubes examined.

1). Fuel Vent Tube in R/H Side of SWB#2 (RBL 98.5)

Conductivity = 40.24 - 40.77 %IACS

Thickness = .045"

Correction factor for thickness = +0.5%

Resulting corrected range = 40.74 - 41.27

Based on these values, it appears that the materials' temper was not affected by heat.

2). Fuel Vent Tube pieces, R/H Wing, Front Spar

- Tube is fractured in two pieces
- Tube has ink marking of 6061-T4 per WW-T-700/6, 028" x 1 3/4" dia.
- Tube was identified as 6SB92232-2: Drawing was checked and verified ink stamp identification of material, heat treat, and thickness.

Conductivity measurements were taken on both pieces in various areas:

Conductivity range = 44.08 - 46.77% IACS

Thickness correction factor = + 215% IACS

Resulting corrected range = 46.58 - 49.27% IACS

Since acceptance range for -T4 material is 36.0 - 45.5% IACS, it appears that the material was artificially aged to a T6 type temper during the accident.

\* This tube is going to the NTSB lab

3) Fuel Vent Tube, R/H Wing, Trailing Edge

- Identified as 1007-4, Tank 4R, 2 1/2" Dia.
- Material thickness = 0.065"

Conductivity range = 41.32 - 43.99% IACS

No thickness correction factor

Material falls within the acceptable conductivity range of -T6 and -T4 therefore it appears the temper of the material was not altered.

4). Air Distribution Ducts - No part number identification

A). .063" thick,

Conductivity range = 40.31 - 41.8% IACS

No thickness correction factor.

B). 065" thick,

Conductivity range = 40.90 - 41.47% IACS  
No thickness correction.

Both distribution ducts exhibit conductivity values which indicate that no temper alterations took place due to temperature exposure during the accident.

JR Strauss 10/2/96

Frank decided to send all three to the  
NTSB Lab - couldn't get in touch with  
Bob Swaim.  
Conductivity Results provided by Boeing Personnel,  
Not independently verified by other  
parties J. Wilby, II  
1/24/97

11.2.1

BARE ALLOYS (Continued)

TABLE I CONDUCTIVITY AND HARDNESS ACCEPTANCE LIMITS (Continued)

ALLOY AND TEMPER		CONDUCTIVITY PERCENT IACS		ROCKWELL HARDNESS			
		MIN.	MAX.	SCALE	THICKNESS RANGE (INCH)	MIN.	MAX.
5386	HXX	---	---	E	0.060 and up	80	---
				15T	0.026 to 0.059	72	---
5456	0	---	---	FL 4	---	---	70
	H112	---	---	FL 4	---	70	---
	H311	---	---	FL 4	---	75	---
	H321	---	---	FL 4	---	90	---
	H323	---	---	FL 4	---	90	---
	H343	---	---	FL 4	---	94	---
6013	T4X	37.0	40.0	B	0.040 and up	41	60
				15T	0.026 to 0.039	74	80
	T6X	41.0	44.0	B	0.040 and up	64	79
				15T	0.026 to 0.039	79.5	86
6061	0	47.0	56.0	E	0.060 and up	---	25
				15T	0.026 to 0.059	---	40.5
	T4XXX	36.0	45.5	E	0.060 and up	64	---
				15T	0.026 to 0.059	63	---
	T42	36.0	45.5	E	0.060 and up	60	---
				15T	0.026 to 0.059	60.5	---
	T6XXX	40.0	51.0	E	0.040 and up	87	90
	FL 5			15T	0.026 to 0.039	78	85.5
6063	0	57.0	65.0	H	---	---	70
	T1X	48.0	58.0	E	0.060 and up	37.0	---
				15T	0.026 to 0.059	47.5	---
	T4X	48.0	58.0	E	0.060 and up	40	---
				15T	0.026 to 0.059	49	---
	T5X	50.0	60.0	E	0.060 and up	44	---
				15T	0.026 to 0.059	51	---
	T6X	50.0	60.0	E	0.060 and up	70	---
				15T	0.026 to 0.059	66	---
7049 FL 6	0	44.0	50.0	E	0.060 and up	---	70
				15T	0.026 to 0.059	---	66
	T73XXX	40.0	44.0	B	0.040 and up	82	---
				15T	0.026 to 0.039	87	---
	T76XXX	38.0	44.0	B	0.040 and up	86	---
			15T	0.026 to 0.039	88.5	---	

BAC

5946

Page 10

5.5.2.1

Bare Thin Gauge (Continued)

TABLE XIII SIGMASCOPE SMP1 - BARE THIN GAUGE

UNCORRECTED CONDUCTIVITY VALUES IN PERCENT IACS	SINGLE THICKNESS (GAUGE), INCHES						
	0.016	0.020	0.025	0.032	0.040	0.050	0.063 AND ABOVE
	CORRECTED CONDUCTIVITY VALUES IN PERCENT IACS						
26.0	16.5	20.0	25.0	28.5	27.5	26.5	26.0
27.0	17.5	21.5	26.5	30.0	28.5	27.5	27.0
28.0	18.0	22.5	28.0	31.5	29.5	28.5	28.0
29.0	19.0	23.5	30.0	32.0	30.5	29.5	29.0
30.0	20.0	25.0	31.5	34.0	31.5	30.5	30.0
31.0	21.0	26.0	32.5	35.0	32.5	31.5	31.0
32.0	22.0	27.5	34.0	35.5	33.5	32.5	32.0
33.0	23.0	29.0	35.5	36.5	34.0	33.5	33.0
34.0	24.0	30.0	36.5	37.5	35.0	34.0	34.0
35.0	25.5	31.5	38.0	38.0	36.0	35.0	35.0
36.0	26.5	33.0	39.5	39.0	37.0	36.0	36.0
37.0	27.5	34.0	41.0	40.0	38.0	37.0	37.0
38.0	28.5	35.5	42.0	41.0	39.0	38.0	38.0
39.0	29.5	37.0	43.5	42.0	40.0	39.0	39.0
40.0	31.0	38.5	45.0	42.5	41.0	40.0	40.0
41.0	32.0	40.0	46.5	43.5	42.0	41.0	41.0
42.0	33.0	41.5	48.0	44.5	43.0	42.0	42.0
43.0	34.0	43.0	49.5	45.5	44.0	43.0	43.0
44.0	35.5	44.5	51.0	46.5	45.0	44.0	44.0
45.0	36.5	46.0	52.5	47.5	45.5	45.0	45.0
46.0	38.0	47.5	54.0	48.5	46.5	46.0	46.0
47.0	39.0	49.0	55.5	49.5	47.5	47.0	47.0
48.0	40.5	50.5	56.5	50.0	48.5	48.0	48.0
49.0	41.5	52.0	57.0	51.0	49.5	49.0	49.0
50.0	43.0	53.0	58.5	52.0	50.5	50.0	50.0
51.0	44.0	54.5	60.0	53.0	51.0	51.0	51.0
52.0	45.5	56.0	60.5	54.0	52.0	52.0	52.0
53.0	47.0	57.0	61.5	55.0	53.0	53.0	53.0
54.0	48.0	58.5	62.5	55.5	54.0	54.0	54.0
55.0	49.5	60.0	63.5	56.5	55.0	55.0	55.0

BSS  
7351  
PAGE 32

CENTER FUEL TANK - LOCATIONS OF POSSIBLE  
SIGNIFICANT\* THERMAL DAMAGE AS INDICATED  
BY ELECTRICAL CONDUCTIVITY SURVEY

\* VARIATION OF MORE THAN 2% IACS WITHIN A PART  
OR %IACS OUTSIDE OF ACCEPTANCE RANGE

2024-T351 LOWER SKIN PANEL	FRAGMENT NUMBER	<sup>SL. 114</sup> S/B 28.5-32.0 % IACS	THERMAL DAMAGE
#5	{ CW-216 } { CW-201 } { CW-221 }	29.4 - 30.6	NO
#4	{ CW-201 } { CW-221 }	29.7 - 33.0	(1)
#3	{ CW-206 } { CW-214 } { CW-218 } { CW-219 } CW-207	28.9 - 29.1	NO
	{ CW-209 } { CW-208 } { CW-201 }	29.1 - 30.0	NO
		29.5 - 30.1	NO
#2	{ CW-205 } { CW-201 }	29.6 - 30.9	NO
#1	{ CW-210 } { CW-212 } { CW-204 } { CW-202 } { CW-203 } { CW-217 }	28.5 - 30.1	NO

(1) MODERATE THERMAL DAMAGE ARTIFICIAL AGING APPEARS TO HAVE  
OCCURRED AT APPROXIMATELY RBL 50 - RBL 90 AND S-15 TO S-18

7075-T651 All except front spar 2024-T351  
All span wise

UPPER SKIN PANELS 7075-T651 S/B 30.0-35.0

PANEL NUMBER	FRAGMENT NUMBERS	% IACS	THERMAL DAMAGE
#3	{ CW-101 } { CW-114 }	31.3 - 31.7	NO
#2	{ CW-102 } { CW-104 } { CW-105 } { CW-122 }	30.7 - 31.9	NO
#1	CW-115 CW-102 CW-103 CW-105	31.5 - 31.9 30.8 - 34.0 30.8 - 32.7 31.3 - 32.0	NO (1) (2) NO

(1) SHARP INCREASE IN % IACS, APPROX. C/L ( $\pm$ ) AND JUST FWD OF SPAN-WISE BEAM #1

(2) POSSIBLY SLIGHT THERMAL DAMAGE, LOCALIZED INCREASE IN % IACS AT RBL 0 TO RBL 10 BETWEEN S-2 AND S-4 AND AT RBL 76 - RBL 98 FROM REAR SPAR TO S-4.

BL 0.00 RIB 7075-T6XX S/B 30.0-35.0

PORTION ATTACHED TO MID-SPAR	31.5% - 33.2%	NO
PORTION ATTACHED TO REAR SPAR	30.3% - 36.2%	YES

REAR SPAR VERTICAL STIFFENERS AT RBL 11 AND LBL 11 \*

P/N 65B10835-1,-3 7075-T73 S/B 38.0 - 42.0%

RBL 11	35.9% - 41.8%	(1)
LBL 11	40.1% - 40.9%	NO

(1) VERY LOCALIZED LOW IACS (35.9%) ON AFT FLANGE APPROXIMATELY 20 INCHES ABOVE LOWER SKIN PANEL

\* THESE STIFFENERS ARE STILL ATTACHED TO THE KEEL BEAM.

## UPPER STRINGERS

7075 - T6511

S/B 30.0 - 35.0%

STRINGER #	LOCATION (%IACS)	THERMAL DAMAGE
1		
2	RBL 11 - RBL 105 (29.7 - 30.2)	NO
3	LBL 100 - RBL 105 (32.9 - 33.1)	NO
4	LBL 100 - RBL 110 (32.6 - 32.9)	NO
5		
6	LBL 25 - RHSOB (28.9 - 32.5)	(1)
7	LBL 25 - RHSOB (28.9 - 32.5)	(1)
SWB 1 8		
9	LXL 44 - RBL 52 (32.2 - 33.0)	(2)
10	LBL 33 - RHSOB (33.1 - 33.5), LBL 98 - LBL 52 (32.4 - 33.2)	(3)
11	LBL 33 - RHSOB (33.1 - 33.5), LBL 98 - LBL 52 (32.4 - 33.2)	(3)
12	LBL 45 - RHSOB (32.4 - 33.5)	NO
13	LBL 40 - RHSOB (31.5 - 33.0)	NO
MS 14		
15		
16	LBL 34 - RBL 50 (26.7 - 29.9)	(4)
17		
18		
SWB 2 19	LBL 34 - LXL 11 (31.4 - 33.2), RBL 76 - RHSOB (33.0%)	(5)
20		
21	RBL 34 - RBL 76 (32.7 - 32.8), RHSOB (32.5)	NO
22	LBL 28 - RBL 34 (32.6 - 32.7), RBL 98 - RHSOB (32.7 - 32.8)	NO
23	LBL 11 - RHSOB (32.2 - 32.4), FEW INCHES BREAK AT ~ RBL 80	NO
24	LBL 11 - RBL 76 (29.5 - 29.8), RBL 90 - RHSOB (27.0 - 29.6)	(6)
25	RBL 11 - RBL 76 (32.2 - 32.5), RBL 80 - RHSOB (28.0 - 32.5)	(7)
26	BLO - RBL 76 (33.3 - 33.5), RBL 80 - RHSOB (28.6 - 33.4)	(8)
27	BLO - RBL 57 (31.3 - 31.4)	NO
SWB 3 28		
29		
30	LBL 76 - RHSOB (32.5 - 33.0)	NO
31	LBL 76 - RHSOB (32.6 - 33.5)	NO
32	LBL 76 - RHSOB (32.7 - 32.9)	NO
33	LBL 76 - RHSOB (28.8 - 29.2)	NO
FS 34		

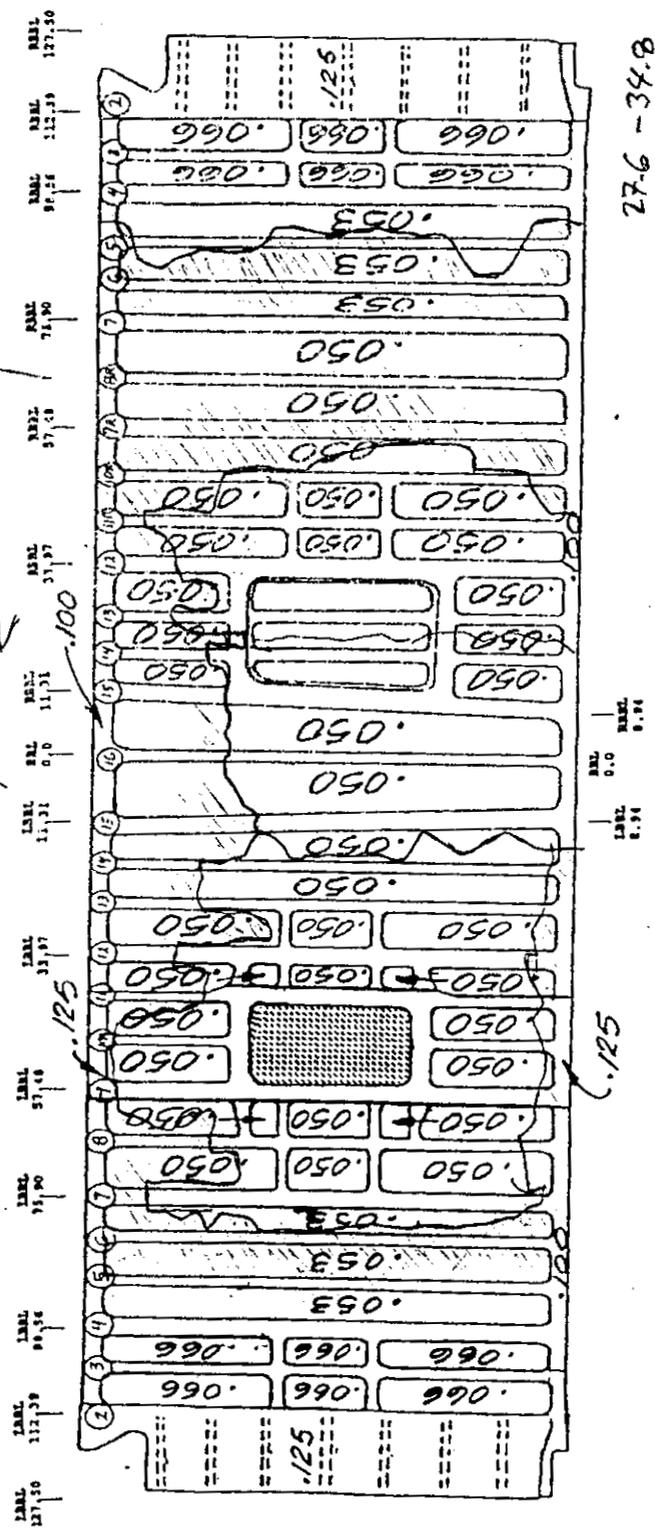
(1) SHARP DROP IN %IACS AT LBL 25 END

(2) BROOM STRAW FRACTURES ON END AT LBL 44

(3) BROOM STRAW FRACTURE ON END AT LBL 33

- (4) ENTIRE PIECE IS BELOW SPEC MINIMUM (SLIGHTLY) BUT WAY LOW FROM APPROXIMATELY LBL 34 - LBL 20, BROOM STRAW FRACTURE AT LBL 34
- (5) SHARP DROP IN IACS FROM LBL 11 (33.2%) TO LBL 34 (31.4%)
- (6) SHARP DROP IN IACS FROM RHS03 (29.6%) TO RBL 90 FRACTURE (27.0%)
- (7) SHARP DROP IN IACS FROM RHS03 (32.5%) TO RBL 80 FRACTURE (28.0%)
- (8) SHARP DROP IN IACS FROM RHS03 (33.4%) TO RBL 80 FRACTURE (28.6%)

31.5 - 31.8  
(CORRECTED PART)



27.6 - 34.8

125

127.50 127.50 112.39 99.54 71.90 57.58 37.97 11.31 0.0 11.31 37.97 57.58 71.90 99.54 112.39 127.50

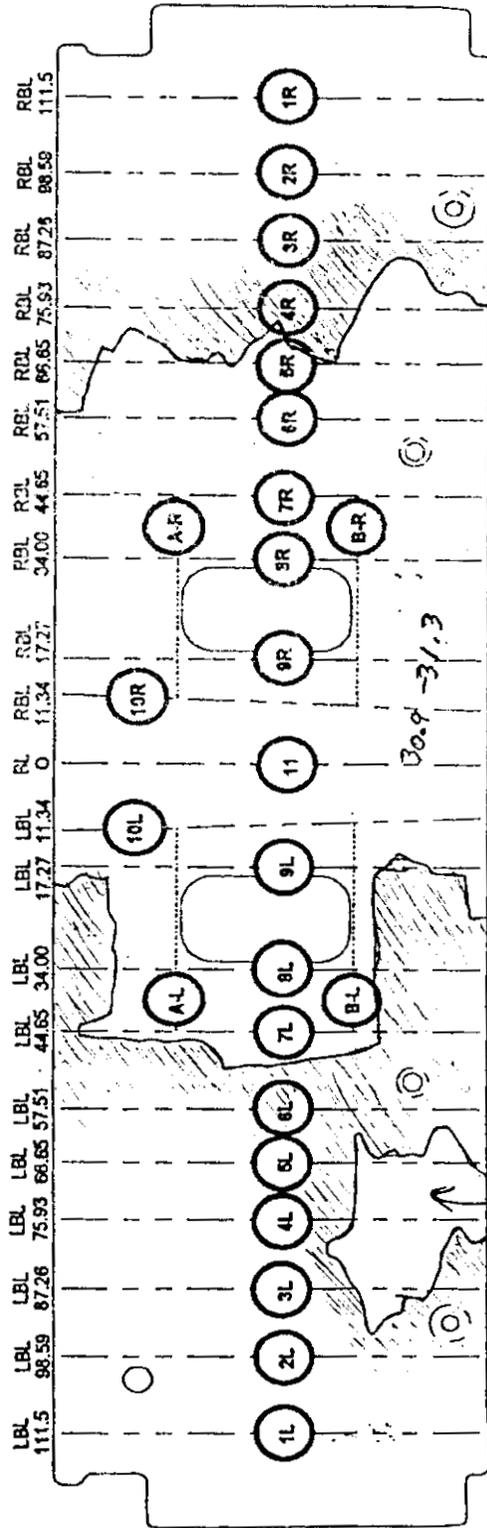
SPANWISE BEAM NO 3.

EMPIRE BEAM NO. 3  
REV. DATE 11/10/77-3  
POOR DATE 4/10/78-1



# 747-100 MID SPAR ASSEMBLY 65B01104-1 INSTL / 65B01104-2 ASSY

**WEB> 65B10279-4, BASIC THICKNESS = .160**



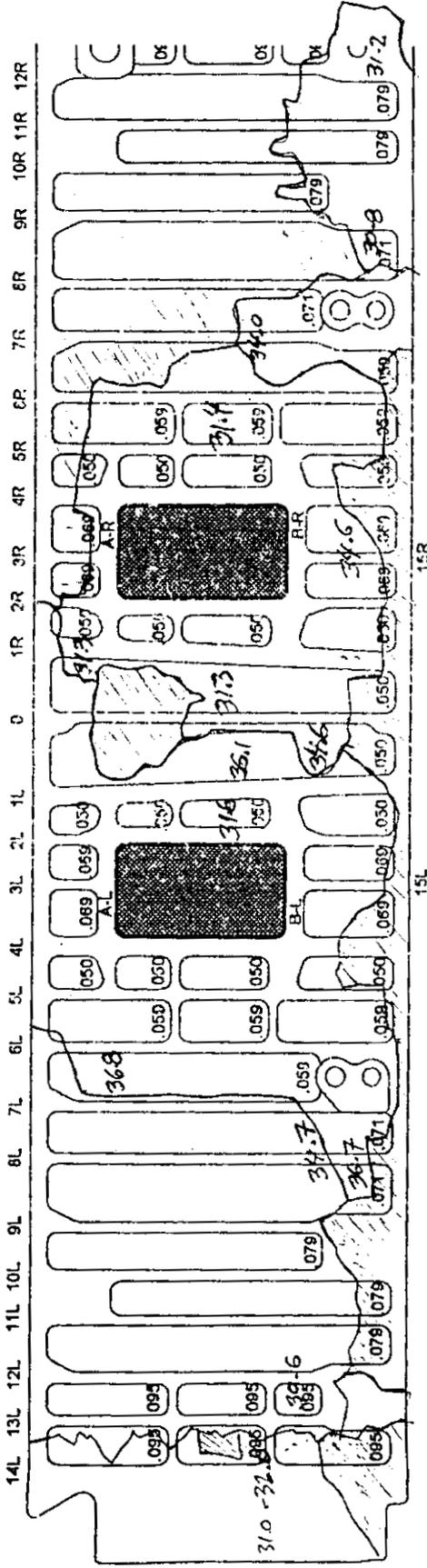
LBL 8.94  
RBL 8.94

31.0-31.4

STIFF NO.	RBL	PART NO	PAD	PAD	EXTRUSION	MATL	PROFILE
1L	111.5 AFT	65B10897			FORGE	7075-773	BATHUB
1L	111.5 FWD	65B10894-4			FORGE	7075-773	BATHUB
2L	98.59 AFT	65B10685-5			BAC1518-446	7178-76511	TRIM - T TO 7 AT UPR 47
3L	87.16 AFT	65B10682-1			BAC1506-2046	7178-76511	CONST - J
4L	75.93 AFT	65B10680-1			BAC1518-446	7178-76511	TRIM - T TO 7 AT UPR 47
5L	64.65 AFT	65B10682-1			BAC1506-2046	7178-76511	CONST - J
6L	53.31 AFT	65B10680-4			BAC1518-446	7178-76511	TRIM - T TO 7 AT UPR 47
7L	44.65 AFT	65B10687-4			BAC1506-2046	7178-76511	CONST - J
8L	34.00 AFT	65B10764-2			BAC1518-446	7178-76511	TRIM - T TO 7 AT UPR 10"
9L	22.77 AFT	65B10687-1			BAC1518-446	7178-76511	TRIM - T TO 7 AT LWR 14" & UPR 10"
10L	8.94 LWR	65B10681-1			BAC1518-448	7178-76511	TRIM - T WITH WIDE FLANGE AT LWR 5" & 7" AT UPR 5"
11	11.34 AFT	65B10813-4			AND10136-1408	7178-76511	CONST - L
A-L	13.44	65B10813-4			AND10136-1408	7178-76511	TRIM - T TO 7 AT BOTH ENDS LAST 5"
B-L	10.41	65B10813-1			AND10136-1408	7178-76511	TRIM - T TO 7 AT BOTH ENDS LAST 5"

# 747-100 SPANWISE BEAM #1 ASSEMBLY 65B01108-1 INSTL / 65B01108-2 ASSY

## 65B10280-4 WEB



STIFF NO.	LBL	PART NO	PAD	EXTRUSION	MATL	PROFILE
0	0.00	65B01108-14	.125	AND10131-1203	7075-T7311	"L" ANGLE
1L	8.64-11.24	65B1076-1	.125	BAC1518-446	7178-T6511	"L" ANGLE
2L	17.27	65B1076-2	.125	BAC1518-446	7178-T6511	"L" ANGLE
3L	25.76	65B1076-3	.125	AND10131-1606	7075-T6511	"L" ANGLE
4L	34.00	65B1076-4	.125	BAC1518-446	7178-T6511	"L" ANGLE
5L	41.77	65B1076-5	.125	BAC1506-2369	7178-T6511	"J" ANGLE
6L	49.60	65B1076-6	.125	BAC1506-2369	7178-T6511	"J" ANGLE
7L	57.31	65B1076-7	.125	BAC1518-446	7178-T6511	"L" ANGLE
8L	66.65	65B1076-8	.125	BAC1518-446	7178-T6511	"L" ANGLE
9L	75.93	65B1076-9	.125	BAC1518-446	7178-T6511	"L" ANGLE
10L	83.24	65B1076-10	.125	BAC1506-2369	7178-T6511	"J" ANGLE
11L	91.14	65B1076-11	.125	BAC1506-2369	7178-T6511	"J" ANGLE
12L	98.59	65B1076-12	.125	BAC1518-446	7178-T6511	"L" ANGLE
13L	106.20	65B1076-13	.125	BAC1506-2369	7178-T6511	"J" ANGLE
14L	112.39	65B1076-14	.125	BAC1506-2369	7178-T6511	"J" ANGLE
15L	123.26	65B1076-15	.125	AND10131-1606	7075-T6511	"L" ANGLE
A-L	HORIZ	65B10813-7	.125	AND10131-2408	7178-T6511	"L" ANGLE
B-L	HORIZ	65B10813-5	.125	AND10131-2408	7178-T6511	"J" ANGLE

STIFF NO.	LBL	FIBEL	PART NO	PAD	EXTRUSION	MATL
1R	8.64-11.24	65B1076-2	.125	BAC1518-446	7178-T6511	
2R	17.27	65B1076-3	.125	BAC1518-446	7178-T6511	
3R	25.76	65B10932-2	.125	AND10131-1606	7075-T6511	
4R	34.00	65B1076-4	.125	BAC1518-446	7178-T6511	
5R	41.77	65B1076-5	.125	BAC1506-2369	7178-T6511	
6R	49.60	65B1076-6	.125	BAC1506-2369	7178-T6511	
7R	57.31	65B1076-7	.125	BAC1518-446	7178-T6511	
8R	66.65	65B1076-8	.125	BAC1506-2369	7178-T6511	
9R	75.93	65B1076-9	.125	BAC1518-446	7178-T6511	
10R	83.24	65B1076-10	.125	BAC1506-2369	7178-T6511	
11R	91.14	65B1076-11	.125	BAC1506-2369	7178-T6511	
12R	98.59	65B1076-12	.125	BAC1518-446	7178-T6511	
13R	106.20	65B1076-13	.125	BAC1506-2369	7178-T6511	
14R	112.39	65B1076-14	.125	BAC1506-2369	7178-T6511	
15R	123.26	65B10932-2	.125	AND10131-1606	7075-T6511	
A-R	HORIZ	65B10813-8	.125	AND10131-2408	7178-T6511	
B-R	HORIZ	65B10813-6	.125	AND10131-2408	7178-T6511	

